





FLOWHEAD

Adjoint Spalart Allmaras

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Content



S.A. Adjoint Turbulence Model

- Implementation
- Validation
- Distributed
- Ext. Aero. Case with adjointSimpleFoam
- **88 New Objective Function**
 - Implementation
 - Testing
 - Distributed



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No Frozen turbulence assumption

$$\frac{\partial \nu_t}{\partial b_m} \neq 0$$

- Adjoint S.A.
 - Adjoint continuity stays the same

$$\frac{\partial u_j}{\partial x_j} = -\frac{\partial F_\Omega}{\partial p}$$

- New term in the adjoint momentum equations

$$-v_j \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) - \frac{\partial}{\partial x_j} \left[(\nu + \nu_t) \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \right] + \frac{\partial q}{\partial x_i} - \widetilde{\nu} \frac{\partial \widetilde{\nu_a}}{\partial x_i} - \frac{\partial}{\partial x_l} \left(e_{jli} e_{jmq} \frac{\mathcal{C}_S}{S} \frac{\partial v_q}{\partial x_m} \widetilde{\nu} \widetilde{\nu_a} \right) = -\frac{\partial F_{\Omega}}{\partial v_i}$$

New equation for adjoint turbulence

$$\begin{split} \frac{\partial \widetilde{\nu_a}}{\partial x_j} v_j + \frac{\partial}{\partial x_j} \left[\left(\nu + \frac{\widetilde{\nu}}{\sigma} \right) \frac{\partial \widetilde{\nu_a}}{\partial x_j} \right] &= \frac{1}{\sigma} \frac{\partial \widetilde{\nu_a}}{\partial x_j} \frac{\partial \widetilde{\nu}}{\partial x_j} + 2 \frac{c_{b2}}{\sigma} \frac{\partial}{\partial x_j} \left(\widetilde{\nu_a} \frac{\partial \widetilde{\nu}}{\partial x_j} \right) + \widetilde{\nu_a} \widetilde{\nu} \, \mathcal{C}_{\widetilde{\nu}}(\widetilde{\nu}, \vec{v}) \\ &+ \frac{\delta \nu_t}{\delta \widetilde{\nu}} \frac{\partial u_i}{\partial x_j} \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right) + \left(-P + D \right) \widetilde{\nu_a} + \frac{\partial F_{\Omega}}{\partial \widetilde{\nu}} \end{split}$$

S.A. Adjoint B.C.

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88 B.C. At Inlet

$$u_{\langle n \rangle} = -\frac{\partial F_{S_I}}{\partial p} \quad , \qquad \mathbf{u}_{\langle t \rangle} = 0 \quad , \qquad \widetilde{\nu_a} = 0$$

- B.C. At Outlet
 - Adjoint pressure

$$q = u_j v_j + u_{\langle n \rangle} v_{\langle n \rangle} + (\nu + \nu_t) \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) n_j n_i + \underbrace{\widetilde{\nu_a} \widetilde{\nu} + \widetilde{\nu_a} \widetilde{\nu} \frac{\mathcal{C}_S}{S} e_{jmq} e_{jli} \frac{\partial v_q}{\partial x_m} n_l n_i}_{OV_{\langle n \rangle}} + \underbrace{\frac{\partial F_{SO}}{\partial v_{\langle n \rangle}}}_{OV_{\langle n \rangle}} e_{jmq} e_{jli} \frac{\partial v_q}{\partial x_m} n_l n_i + \underbrace{\frac{\partial F_{SO}}{\partial v_{\langle n \rangle}}}_{OV_{\langle n \rangle}} e_{jmq} e_{jli} \frac{\partial v_q}{\partial x_m} n_l n_i + \underbrace{\frac{\partial F_{SO}}{\partial v_{\langle n \rangle}}}_{OV_{\langle n \rangle}} e_{jmq} e_{jli} \frac{\partial v_q}{\partial x_m} n_l n_i + \underbrace{\frac{\partial F_{SO}}{\partial v_{\langle n \rangle}}}_{OV_{\langle n \rangle}} e_{jmq} e$$

- Adjoint velocity

$$0 = \mathbf{u}_{\langle t \rangle} v_{\langle n \rangle} + (\nu + \nu_t) \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) n_j t_i \underbrace{ \underbrace{ \tilde{\nu_a} \tilde{\nu} \frac{\mathcal{C}_S}{S} c_{jmq} c_{jli} \frac{\partial v_q}{\partial x_m} n_l t_i}_{S} + \frac{\partial F_{SO}}{\partial \mathbf{v}_{\langle t \rangle}} \underbrace{ \frac{\partial F_{SO}}{\partial \mathbf{v}_{\langle t \rangle}}}_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_i }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_q}{\partial x_m} \right) n_j t_j }_{S} \underbrace{ \left(\frac{\partial v_q}{\partial x_m} + \frac{\partial v_$$

Adjoint turbulence field

$$0 = -\frac{\delta\nu_t}{\delta\widetilde{\nu}}u_i \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i}\right)n_j + \widetilde{\nu_a}v_j n_j + \left(\nu + \frac{\widetilde{\nu}}{\sigma}\right)\frac{\partial\widetilde{\nu_a}}{\partial x_j}n_j + \frac{\partial F_{S_O}}{\partial\widetilde{\nu}}$$

B.C. At Walls

$$u_{\langle n \rangle} = -\frac{\partial F_{S_W}}{\partial p}$$
, $\mathbf{u}_{\langle t \rangle} = 0$, $\widetilde{\nu_a} = 0$

S.A. Adjoint Sensitivity

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Sensitivity calculation

$$\frac{\delta J}{\delta b_n} = -\int_{\Gamma_W} \left[\nu \frac{\partial u_i}{\partial x_j} n_j - q n_i \right] \frac{\partial v_i}{\partial x_k} \frac{\delta x_k}{\delta b_n} d\Gamma - \int_{\Gamma_W} \nu \frac{\partial \widetilde{\nu_a}}{\partial x_j} n_j \frac{\partial \widetilde{\nu}}{\partial x_k} \frac{\delta x_k}{\delta b_n} d\Gamma + \int_{\Omega} \widetilde{\nu_a} \widetilde{\nu} \, \mathcal{C}_d(\widetilde{\nu}, \vec{v}) \frac{\partial d}{\partial b_n} d\Omega$$

Out of all the terms which have been shown to be negligible, this term is the most costly.

Zymaris A, Papadimitriou D, Giannakoglou K., Othmer C. Continuous Adjoint Approach to the Spalart-Allmaras Turbulence Model for Incompressible Flows. *Computers & Fluids*, 38(8):1528–1538, 2009.

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• Validation is tedious through finite differencing.

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 Slight modification to the primal SA to obtain "correct" F.D.

• Wrongly signed derivatives with frozen turbulence !

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- Elongated S-Bend Duct case
- 2 million cells
- ₩ Y+ < 1 everywhere
- Re = 16000
- Power Dissipation Objective

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Good Convergence Obtained – Not without any effort !



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Adjoint Turbulence Model Usage

adjointTurbulen	ceModel	adjointSpalartAllmaras;					
turbulence	on;						
adjointLaminarCo	oeffs						

adjointSpalartAllmarasCoeffs

alphaNut	alpł	າaNເ	ıt	[() () () (0 (0 0] 1.5
Cbl	Cbl	[0]	0	0	0	0	0	0]	0.1355;
Cb2	Cb2	[0]	0	0	0	0	0	0]	0.622;
Cw2	Cw2	[0]	0	0	0	0	0	0]	0.3;
Cw3	Cw3	[0]	0	0	0	0	0	0]	2;
Cvl	Cv1	[0]	0	0	0	0	0	0]	7.1;
Cv2	Cv2	[0]	0	0	0	0	0	0]	5.0;

dimensions internalField boundaryField { inlet {	[0 0 -1 uniform	0 0 0 0]; 0;
type		fixedValue;
value		uniform 0;
}		
outlet		
{		
type		adjointOutletNuaTilda;
value		uniform 0;
}		
wall		
{		
type		fixedValue;
value		uniform 0;
}		
}		

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Additional equation and B.C. for nuaTilda.

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- Additional terms in the adjoint momentum equation.
- Adjoint pressure equation and B.C. are unaffected.
- Adjoint SA model activated through constant/adjointTurbulenceProperties file.

External Aero case

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- **B6 test case**
- Steady RANS calculation
- foamProMesh:~5m cells 3 surface layers (less than normal)
- Surface mesh was priority!



External Aero case

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- Surface sensitivities for Drag minimisation
- Special interest in the rear of the car and the mirror
- Why Not S.A. Adjoint ?



In general the obvious cost function to directly optimise wrt turbulence is

$$J = \int_{V} k dV$$

However, the Spalart-Allmaras model was chosen for adjointing because of its (relative!) simplicity. This doesn't give k directly.

Adjoint of k-epsilon model is one of the next steps

Turbulence based cost function



- We needed to construct a suitable equivalent.
- The only turbulence information directly available is $\widetilde{\mathcal{V}}$, which we need to drive towards zero
- So the obvious cost function is $\int_{V} C \tilde{v}^2 dV$ where C is an arbitrary constant scaling factor V
- In practice we defined $C = \frac{1}{d^2}$ where d is interpreted as a turbulence length scale.
- This gives us dimensional consistency with the original idea of using turbulent kinetic energy.

Turbulence based cost function

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SIMPLE { nNonOrthogonalCorrectors 0; solvePrimal yes; solveAdjoint no; adjointTurbulence yes; ... turbEnergyOptimisation yes;

turbEnergyLengthScale 0.001;

 Additional options in the system/fvSolution file

- Gw 7 6 4 2 0 -2 -3
- Example of sensitivity field obtained using this cost function.



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Low-Re Adjoint S.A. Implemented – High constraints on mesh quality

New Cost function for turbulence minimization

Requirement for High-Re adjoint turbulence model



For any information or inquiry please contact: flowhead@iconcfd.com

Special thank you to NTUA





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