

Continuous and Discrete adjoint methodologies within ESI CFD solvers

Guillaume Pierrot ESI Group





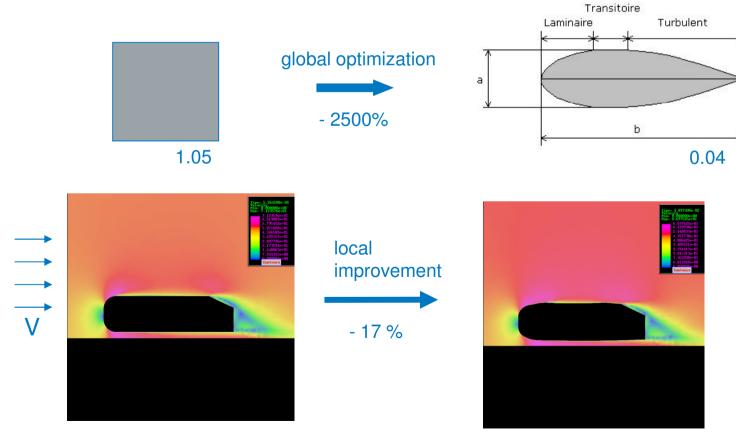


- Continuous Adjoint Solver within PAM-FLOW
- Discrete Adjoint solver interfaced with CFD-ACE+
- Morphing



Rationale

Different problematics:



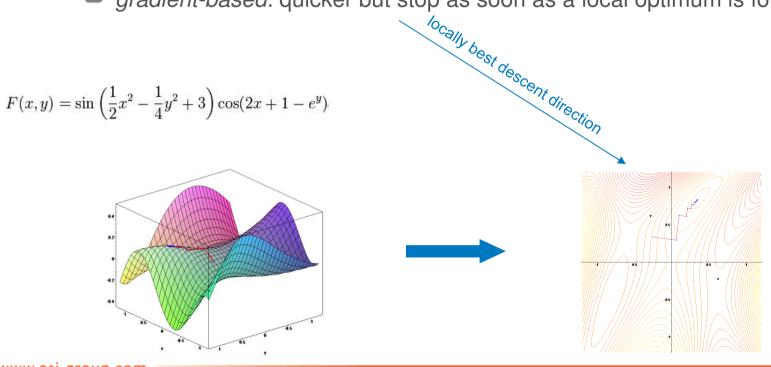
Rationale

- One would like to improve the performance of some given design wrt some criterion: the "cost function"
 - may be lift, drag..
 - design parameters may be nodes coordinates, CAD parameters, level set position..
- Standard numerical simulation provides a way to evaluate a design "a posteriori"
- Optimal design tools automatically select the best (or at least a better) design wrt some given criterion

Rationale

Different optimization methods:

- non-gradient based (e.g. genetic algorithms): slow but may succed in finding a global optimum
- gradient-based: quicker but stop as soon as a local optimum is found







Different approaches for computing the gradient:

- by Finite Differences → PAM-OPT
- by using the adjoint state \rightarrow PAM-FLOW Adjoint Solver, i-adjoint

Both have their pro and cons

Rationale

By Finite Differences:

- Flexible, black box tool: very generic, no knowledge on the underlying application solver is needed
- But requires a number of runs proportional to the number of design parameters
- Not sustainable when it tends to be large (e.g. free shape optimization)
- In pratic, used together with a surrogate model (for CPU savings), which introduces further approximation and complexity

$$\nabla_{\beta} I \approx \left(\frac{I_{\beta_{1}+\delta\beta_{1}}-I_{\beta_{1}}}{\delta\beta_{1}}, \frac{I_{\beta_{2}+\delta\beta_{2}}-I_{\beta_{2}}}{\delta\beta_{2}}, ..., \frac{I_{\beta_{n}+\delta\beta_{n}}-I_{\beta_{n}}}{\delta\beta_{n}} \right)$$



Rationale

By using the adjoint state:

- Less generic: does require some knowledge of the underlying application code
- More complicated → requires a dedicated tool, the so-called « adjoint solver »
- But requires only one primal+one adjoint run → cost is independant on the number of design parameters
- Well suited for shape optimization





2 different approaches for computing the adjoint state:

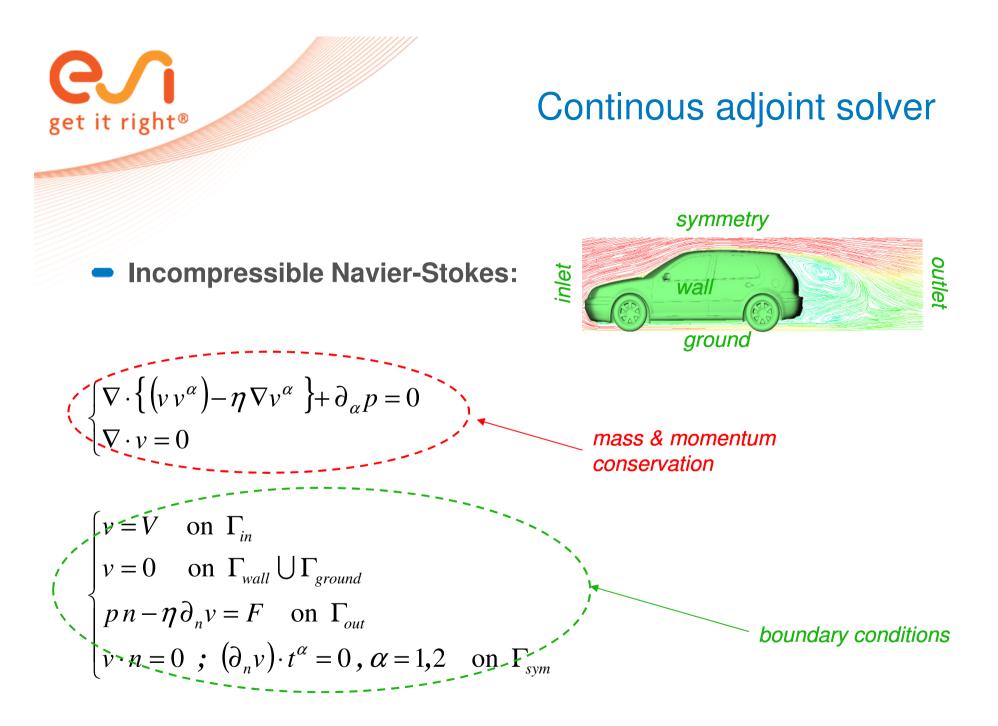
- Linearize/Dualize then Discretize → Continuous Adjoint Method
- Discretize then Linearize/Dualize → Discrete Adjoint Method



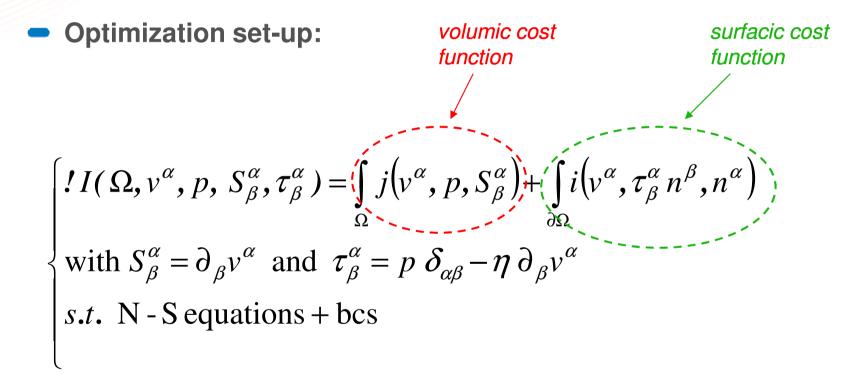


ESI adjoint solutions:

- Continuous adjoint solver embedded into PAM-FLOW (vertex-centered FV) (2006)
- Discrete adjoint library interfaced with CFD-ACE+ (cell-centered FV,multiphysics) (2012)



Continous adjoint solver



Continous adjoint solver

KKT theory:

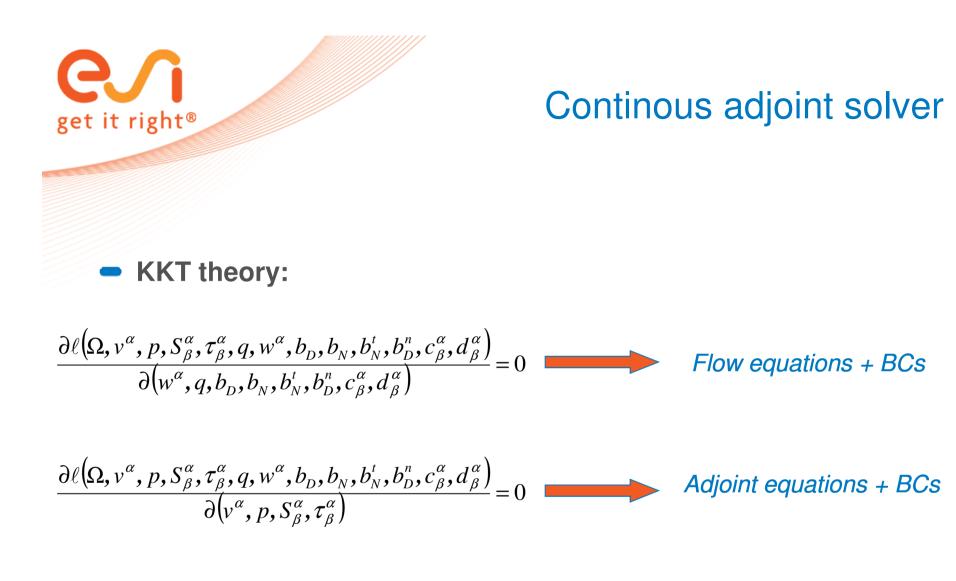
*flow eqs bcs*constitutive relations

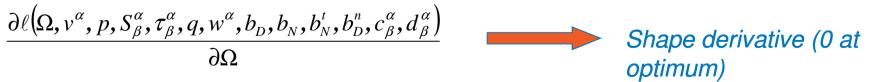
 $\ell\left(\Omega, v^{\alpha}, p, S^{\alpha}_{\beta}, \tau^{\alpha}_{\beta}, w^{\alpha}, q, b_{N}, b_{D}, b_{N}^{t}, b_{D}^{n}, c^{\alpha}_{\beta}, d^{\alpha}_{\beta}\right) =$

$$I(\Omega, v^{\alpha}, p, S^{\alpha}_{\beta}, \tau^{\alpha}_{\beta}, w^{\alpha}, q) + \int_{\Omega} \nabla \cdot \left\{ (v v^{\alpha}) + (\tau^{\alpha}_{\beta})_{\beta} \right\} w^{\alpha}$$

$$\left(- \int_{\Omega} q \nabla \cdot v \right) + \int_{\Gamma_{out}} \left\{ F - (\tau^{\alpha} \cdot n)_{\alpha} \right\} b_{N} + \int_{\Gamma_{in} \cup \Gamma_{wall} \cup \Gamma_{ground}} (v - \delta_{in} V) b_{D} + \int_{\Gamma_{out}} (\tau^{\alpha} \cdot n) t^{\alpha} b_{N}^{t} + \int_{\Gamma_{sym}} (v \cdot n) b_{D}^{N} + \int_{\Omega} \left\{ \overline{\tau^{\alpha}_{\beta}} - (p \delta_{\alpha\beta} - \eta \partial_{\beta} v^{\alpha}) \right\} c^{\alpha}_{\beta}$$

$$+ \int_{\Omega} \left(S^{\alpha}_{\beta} - \partial_{\beta} v^{\alpha} \right) d^{\alpha}_{\beta_{i}}$$

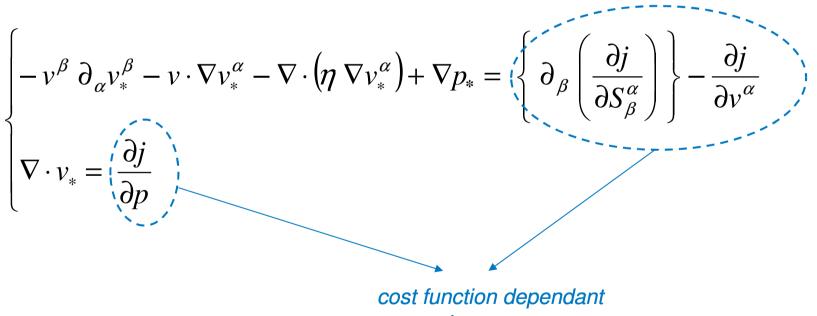






Continous adjoint solver

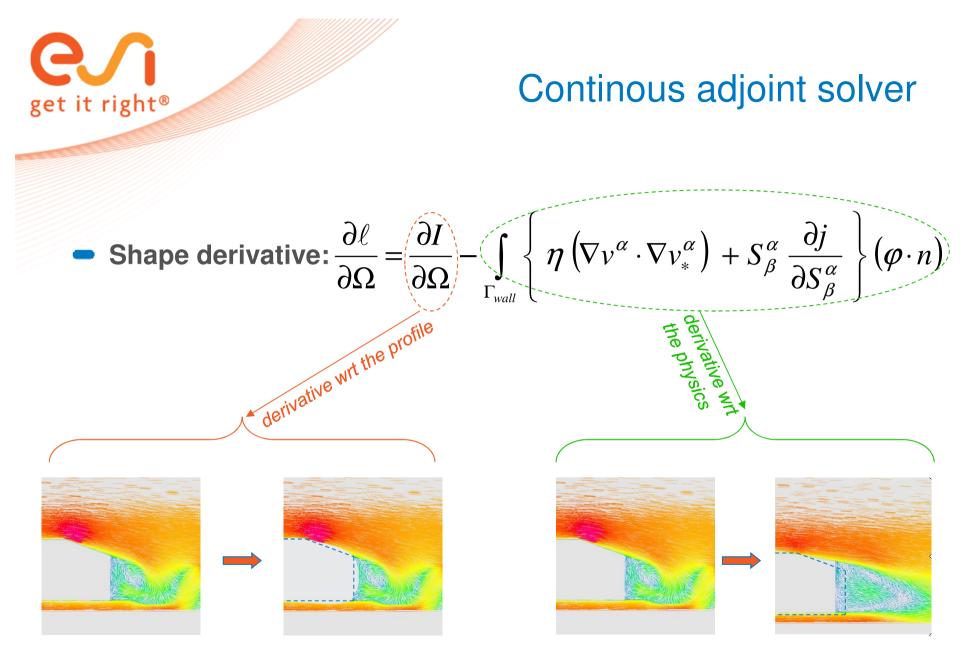
Adjoint equations:



source terms

Continous adjoint solver

• Adjoint bcs: $\begin{cases}
v_* = \left(-\frac{\partial i}{\partial (\tau^{\alpha} \cdot n)} \right) \text{ on } \Gamma_{in} \cup \Gamma_{wall} \cup \Gamma_{ground} \\
v_* = \left(-\frac{\partial i}{\partial (\tau^{\alpha} \cdot n)} \right) \text{ on } \Gamma_{in} \cup \Gamma_{wall} \cup \Gamma_{ground} \\
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v_* = \left(-\frac{\partial i}{\partial (\tau^{\alpha} \cdot n)} \cdot n \right); \quad (\eta \partial_n v_*) \cdot t^{\alpha} = \left(-\frac{\partial i}{\partial v} \cdot t^{\alpha} - \left(-\frac{\partial j}{\partial S_{\beta}^{\alpha}} n^{\beta} \right) \right) \text{ on } \Gamma_{out} \\
v_* = \left(-\frac{\partial i}{\partial (\tau^{\alpha} \cdot n)} \cdot n \right); \quad (\eta \partial_n v_*) \cdot t^{\alpha} = \left(-\frac{\partial i}{\partial v} \cdot t^{\alpha} - \left(-\frac{\partial j}{\partial S_{\beta}^{\alpha}} n^{\beta} \right) \right) \text{ on } \Gamma_{sym}$



keep the flow / change the shape

keep the shape / change the flow

Continous adjoint solver get it right® cost function **Application to Aero Force:** $I = \int_{\Gamma_{wall}} \{ p(n \cdot d) - \eta \partial_n (v \cdot d) \} = \int_{\Gamma_{wall}} (\tau^{\alpha} \cdot n)_{\beta} \cdot d$ $\begin{bmatrix} -v^{\beta} \partial_{\alpha} v_{*}^{\beta} - v \cdot \nabla v_{*}^{\alpha} - \nabla \cdot (\eta \nabla v_{*}^{\alpha}) + \nabla p_{*} = 0 \\ \nabla \cdot v_{*} = 0 \end{bmatrix}$ adjoint system $\frac{\partial \ell}{\partial \Omega} = -\int_{\Gamma_{wall}} \eta \left(\nabla v^{\alpha} \cdot \nabla v_{*}^{\alpha} \right) (\varphi \cdot n)$ source term $\begin{cases} v_* = (-d) \text{ on } \Gamma_{wall} \\ v_* = 0 \text{ on } \Gamma_{in} \cup \Gamma_{ground} \\ p_* n - \eta \partial_n v_* = (v \cdot v_*) n + (v \cdot n) v_* \text{ on } \Gamma_{out} \\ v_* \cdot n = 0 \text{ ; } (\eta \partial_n v_*) \cdot t^{\alpha} = 0, \alpha = 1,2 \text{ on } \Gamma_{sym} \end{cases}$ shape derivative adjoint bcs

Continous adjoint solver get it right®

• **Pressure Drop:** $I = + \int_{\Omega} \eta (\nabla v^{\alpha})^2 = - \int_{\Gamma_{in} \cup \Gamma_{out}} \{p + v^2/2\} (v \cdot n) - \eta \partial_n v^2 = - \int_{\Gamma_{wall}} v^2/2 (v \cdot n) + (\tau^{\alpha} \cdot n)_{\beta} \cdot v$

$$\begin{cases} -v^{\beta} \partial_{\alpha} v_{*}^{\beta} - v \cdot \nabla v_{*}^{\alpha} - \nabla \cdot (\eta \nabla v_{*}^{\alpha}) + \nabla p_{*} = 0 \\ \nabla \cdot v_{*} = 0 \end{cases}$$

$$adjoint system$$

$$\begin{cases} v_{*} = 0 \quad \text{on } \Gamma_{wall} \quad \text{source terms} \\ v_{*} = -V \quad \text{on } \Gamma_{in} \\ p_{*} n - \eta \partial_{n} v_{*} = (v \cdot v_{*}) n + (v \cdot n) v_{*} \\ - (v^{2}/2)n - v (v \cdot n) \\ -F \quad \text{on } \Gamma_{out} \end{cases}$$

$$adjoint bcs$$

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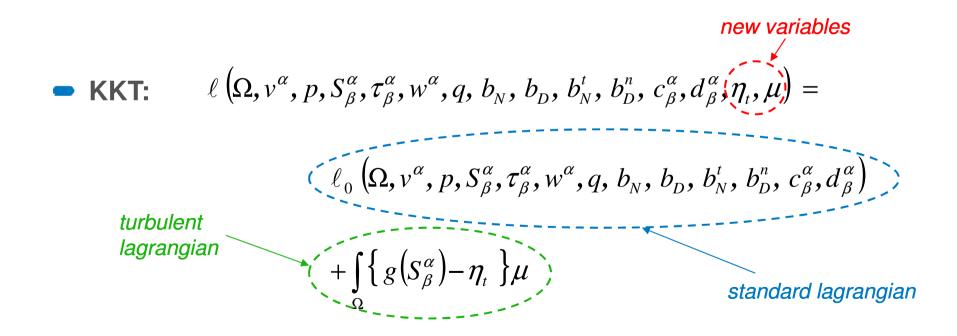
cost function

Continous adjoint solver

Some complication: turbulent viscosity

$$\begin{cases} \eta = \eta + \eta_t \\ \eta_t = g(S^{\alpha}_{\beta}) \end{cases}$$

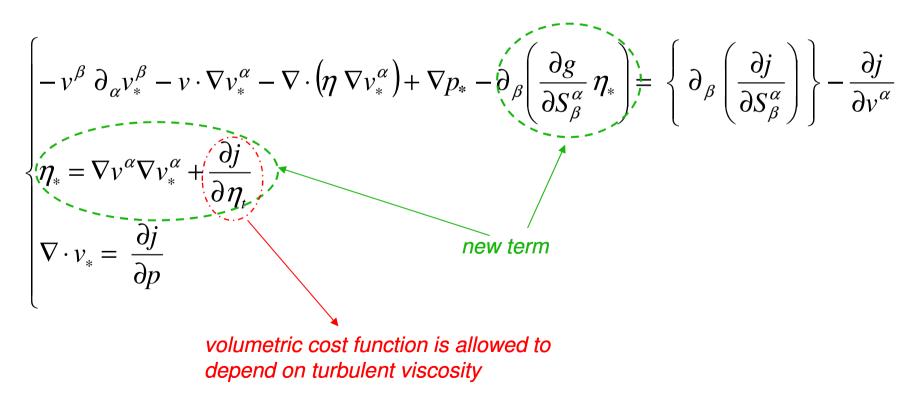
Smagorinsky model





Continous adjoint solver

Modified adjoint equations:

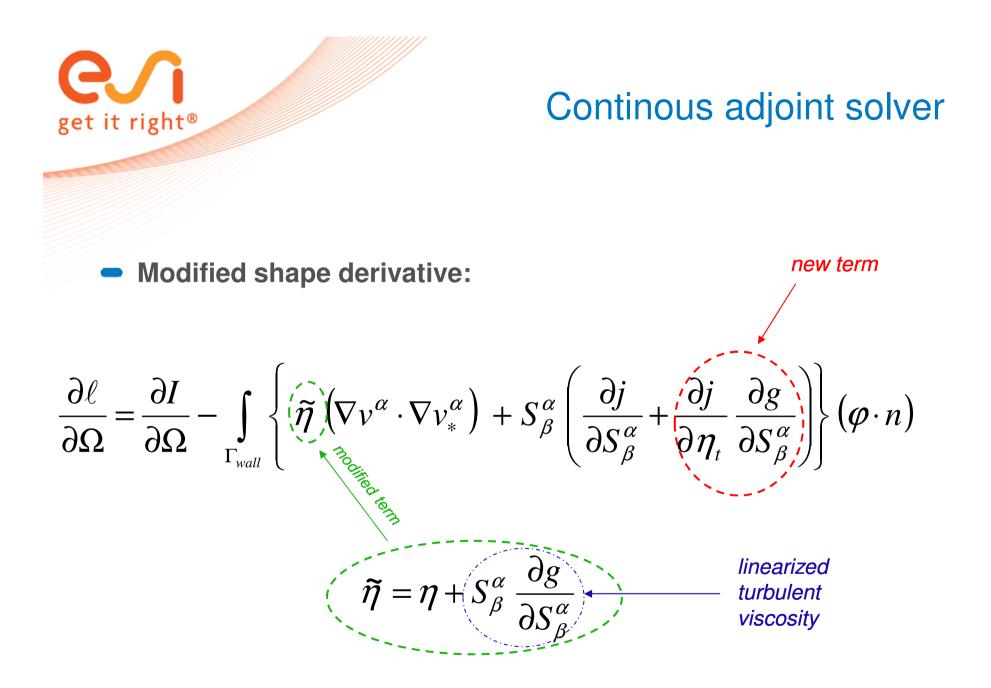




Continous adjoint solver

Modified adjoint bcs:

$$\begin{cases} v_* = -\frac{\partial i}{\partial (\tau^{\alpha} \cdot n)} & \text{on } \Gamma_{in} \cup \Gamma_{wall} \\ p_* n - \eta \partial_n v_* = (v \cdot v_*) n + (v \cdot n) v_* + \frac{\partial i}{\partial v} + \left(\frac{\partial j}{\partial S^{\alpha}_{\beta}} n^{\beta}\right)_{\alpha} (+ \left(\frac{\partial g}{\partial S^{\alpha}_{\beta}} n^{\beta}\right) \eta_*) & \text{on } \Gamma_{out} \\ v_* \cdot n = -\frac{\partial i}{\partial (\tau^{\alpha} \cdot n)} \cdot n \quad ; \quad (\eta \partial_n v_*) \cdot t^{\alpha} = -\frac{\partial i}{\partial v} \cdot t^{\alpha} - \left(\frac{\partial j}{\partial S^{\alpha}_{\beta}} n^{\beta}\right)_{\alpha} \cdot t^{\alpha} + t^{\alpha} \cdot \left(\frac{\partial g}{\partial S^{\alpha}_{\beta}} n^{\beta}\right) \eta_*) \alpha = 1,2 \quad \text{on } \Gamma_{sym} \end{cases}$$



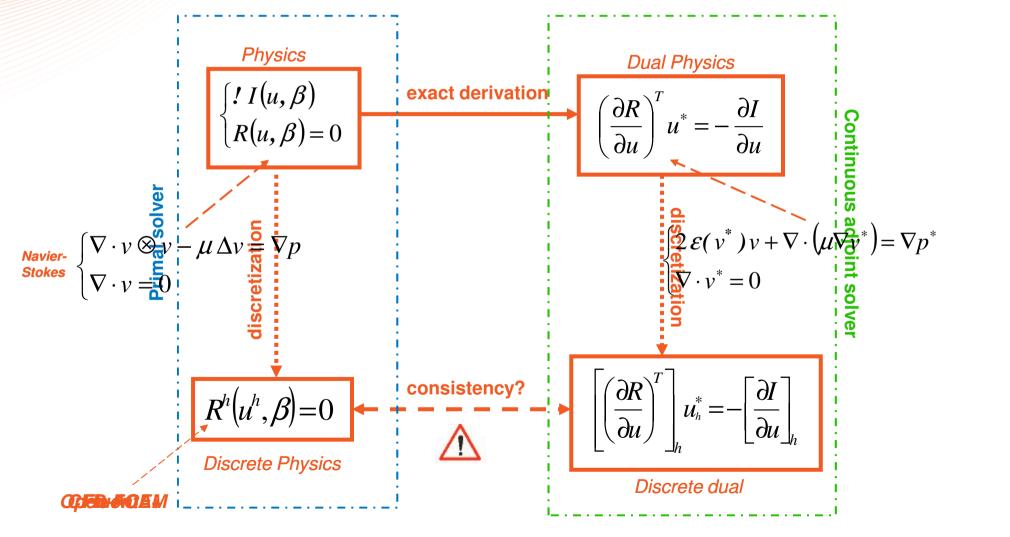
Continous adjoint solver

More complications..

- turbulence models (k-epsilon,Spalart-Allmaras..)
- law of the wall
- natural convection (+Temperature equation)
- MHD
- Chemistry ..



Continous adjoint solver



Continuous adjoint solver

Pros:

- Relatively easy-to-implement
- Independant of the numerical scheme of the application code
- CPU and memory efficient

Cons:

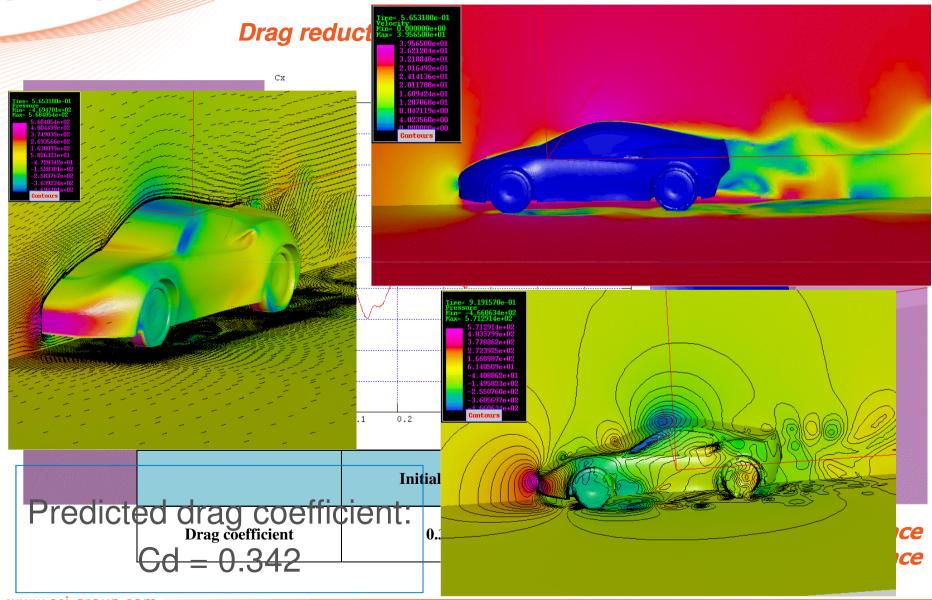
- Gradient inconsistency: the computed adjoint state is not the adjoint state of the computed physical field
- Requires by hand differentiation of the underlying physical model → may be tricky (turbulence models..)
- High cost maintenance: any model addendum in the primal solver requires a specific development effort counterpart in the adjoint solver

Continuous adjoint solver

- Enriched with converters in 2010 so that it can accomodate results of alternative CFD codes
 - the CFD may be run using OpenFOAM or Star-CCM+ and the adjoint using PAM-FLOW
- Both academic and industrial proof of value
- Integrated within ESI Visual process environment

Only tet mesh

Continuous adjoint solver

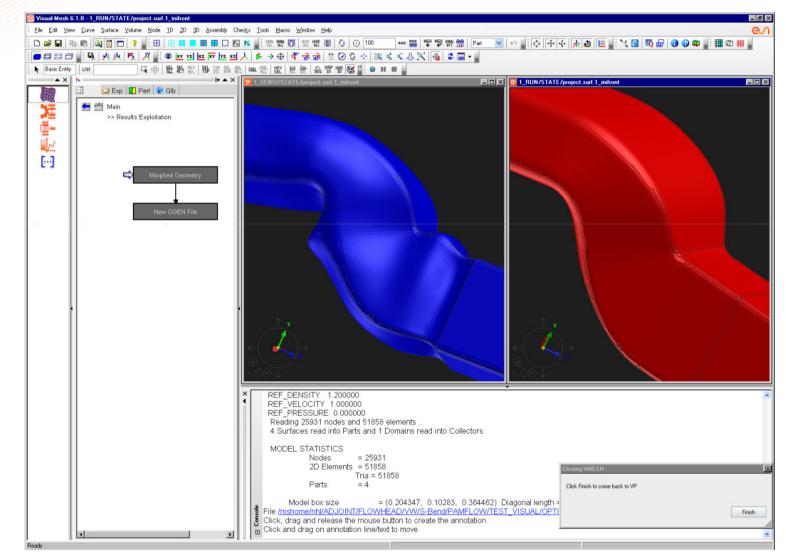


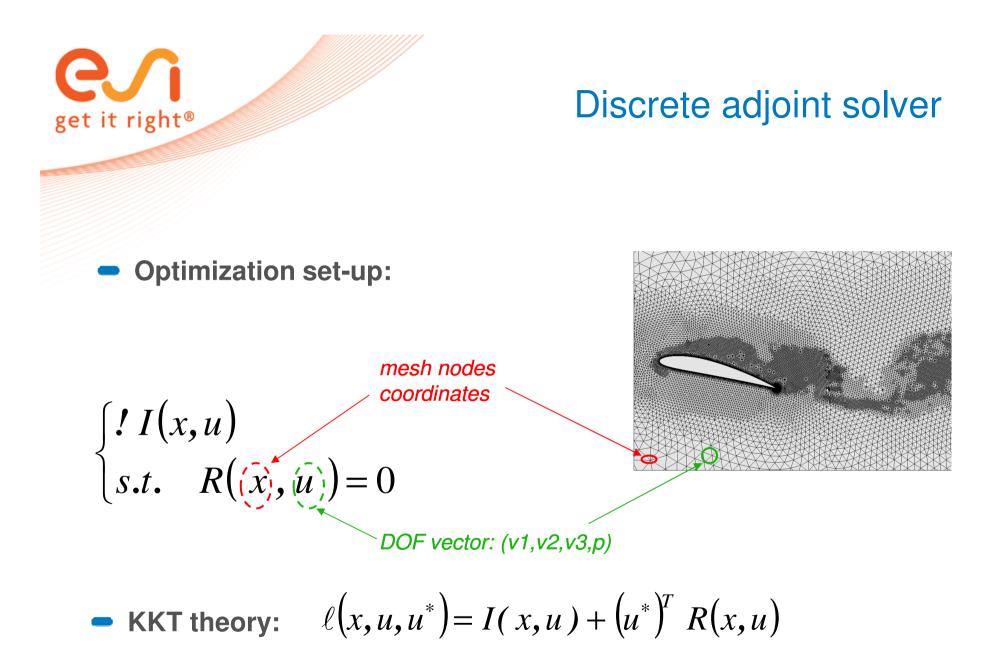
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Continuous adjoint solver



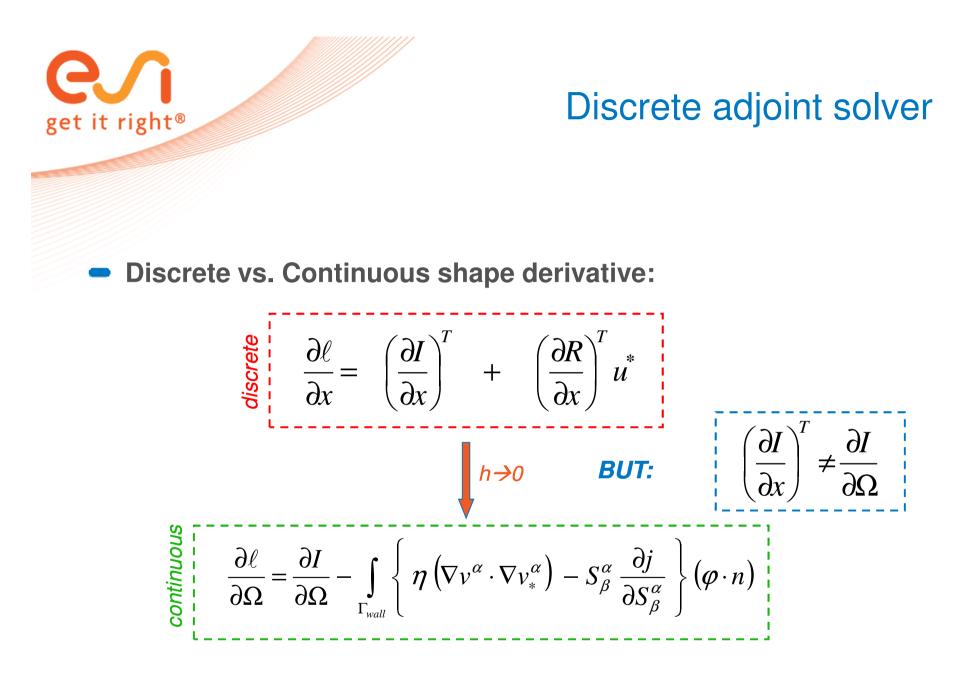


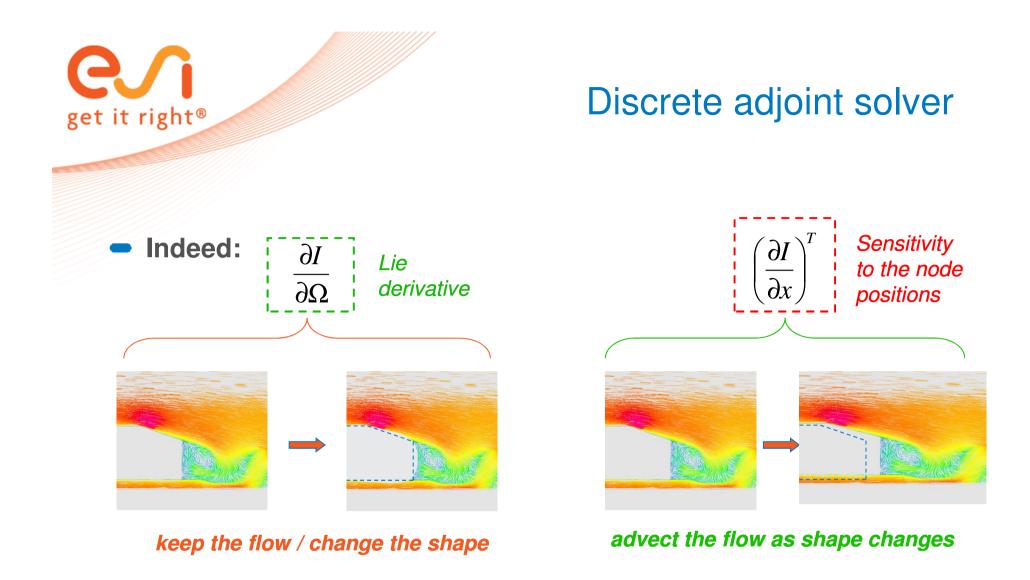


Discrete adjoint solver

• KKT theory:

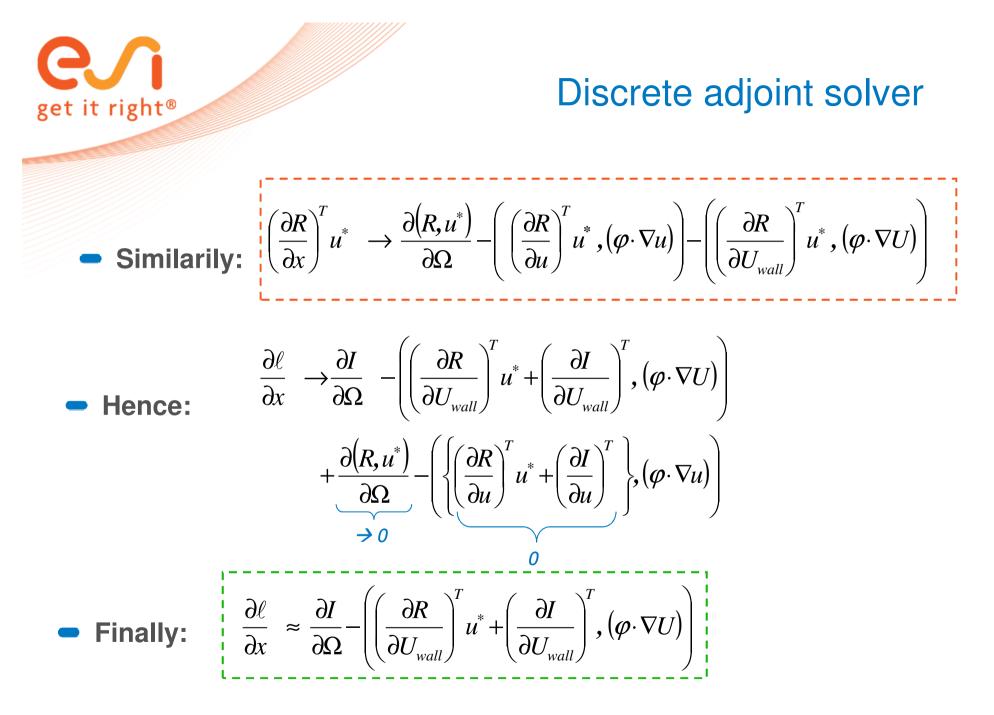
$$\frac{\partial \ell(x, u, u^*)}{\partial u} = \frac{\partial I(x, u)}{\partial u} + \left(\frac{\partial R(x, u)}{\partial u}\right)^T u^* = 0 \quad \longrightarrow \quad \begin{array}{c} \text{discrete} \\ \text{adjoint} \\ \text{equation} \end{array}$$
$$\frac{\partial \ell(x, u, u^*)}{\partial x} = \frac{\partial I(x, u)}{\partial x} + \left(\frac{\partial R(x, u)}{\partial x}\right)^T u^* \quad \longrightarrow \quad \begin{array}{c} \text{sensitivity} \\ \text{vector} \end{array}$$

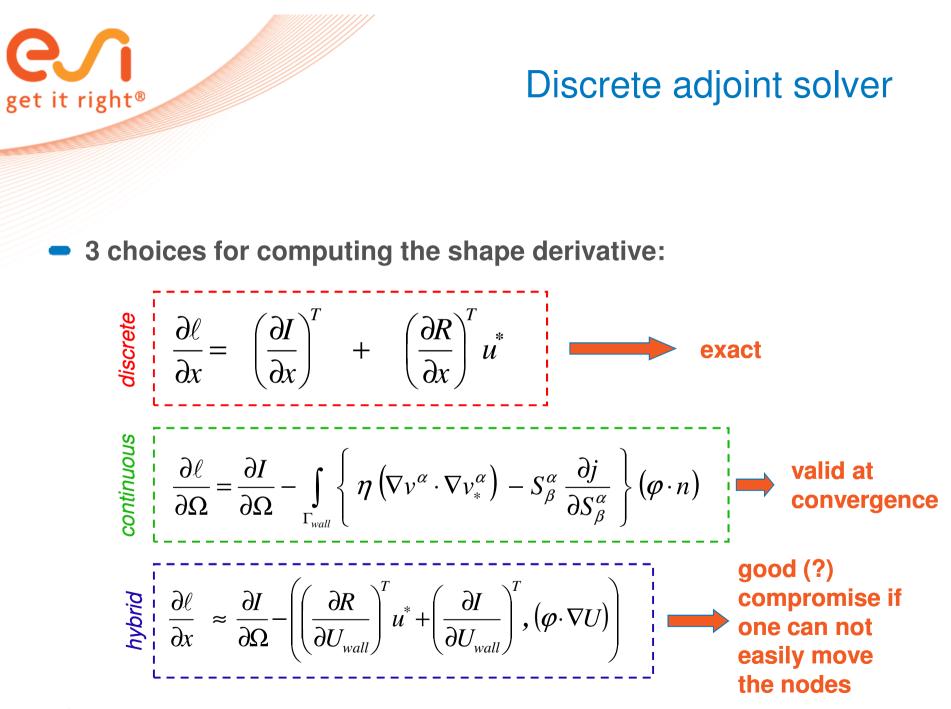




They are connected:

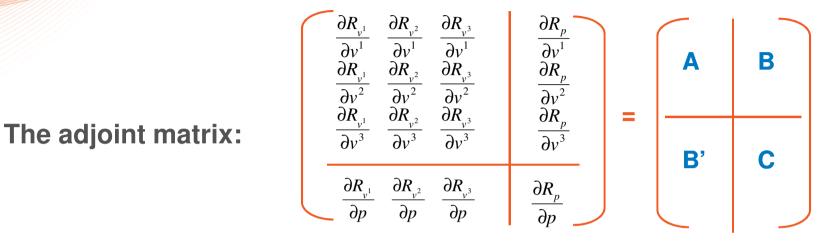
 $\left(\frac{\partial I}{\partial x}\right)^T \rightarrow \frac{\partial I}{\partial \Omega} - \left(\frac{\partial I}{\partial u}\right)(\varphi \cdot \nabla u) - \left(\frac{\partial I}{\partial U_{wall}}\right)(\varphi \cdot \nabla U)$



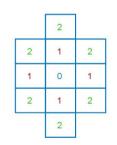


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Discrete adjoint solver



Typical block stencil:



- Huge, (not so) sparse, unsymmetric, undefinite matrix

Hard to solve (saddle-point system)

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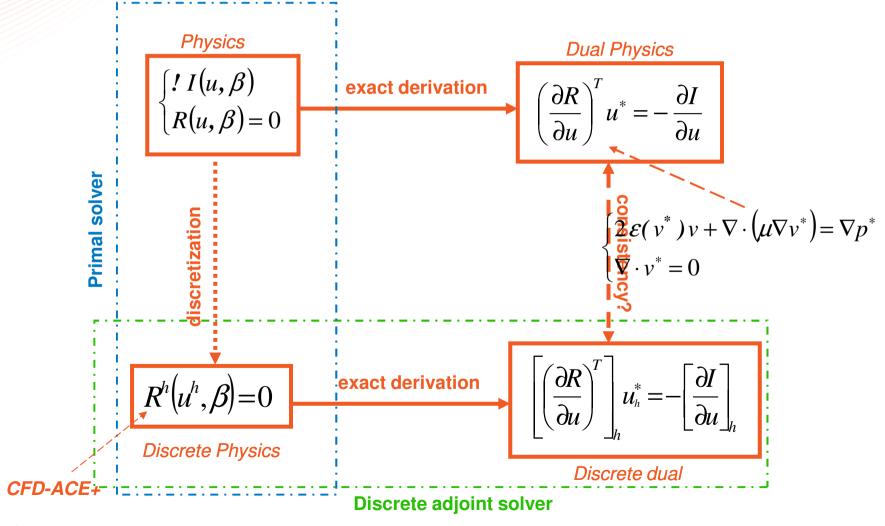
Use an optimal preconditioner (DDM, Multigrid, Subdomain deflation)

- Or a segregated-like approach:
$$P(x^{n+1}-x^n) = \left(\frac{\partial I}{\partial u}\right)^T - \left(\frac{\partial R}{\partial u}\right)^T x^n$$

$$P^{-1} = \begin{pmatrix} I & -D_A^{-1}B \\ 0 & I \end{pmatrix} \begin{pmatrix} I & 0 \\ 0 & (B'D_A^{-1}B - C)^{-1} \end{pmatrix} \begin{pmatrix} I & 0 \\ B' & -I \end{pmatrix} \begin{pmatrix} (D_A + A)^{-1} & 0 \\ 0 & I \end{pmatrix}$$
$$D_A = \alpha \text{ DIAG} (A)$$

- If matrix has to be assembled, out of coring may be needed





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 Gradient consistency: the computed adjoint state is the real adjoint of the computed physical field

Cons:

- Depends on the very numerical scheme of the application code
- How to build the discrete adjoint operator?

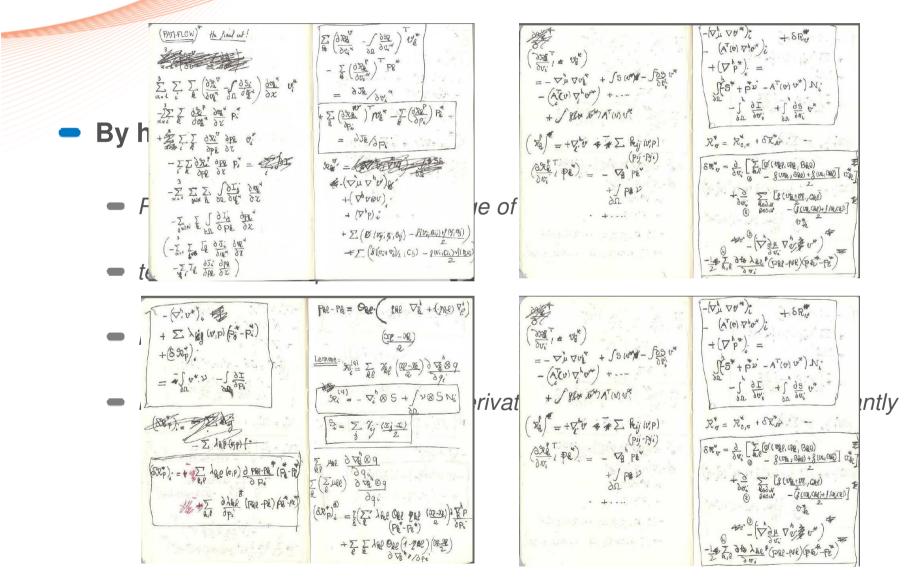


• Different approaches for building the discrete adjoint operator:

- By hand
- Automatic (Algorithmic) Differentiation

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Discrete adjoint solver



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Discrete adjoint solver

By Algorithmic Differentiation:

- The source code istelf is differentiated
- 2 modes: direct / reverse
- 2 approaches: source transformation / operator overloading

Pros & cons:

- Low cost maintenance: the code is differentiated once and for all (no further effort for accomodating newly developped models in the primal solver)
- But each application solver adjoint derivation has to be adressed mostly independently
- May turn to be tedious and time consuming depending on the code structure and programming language
- Very invasive: requires full access to the source code
- Severe CPU efficiency and memory consumption challenges



Academic validation against FD

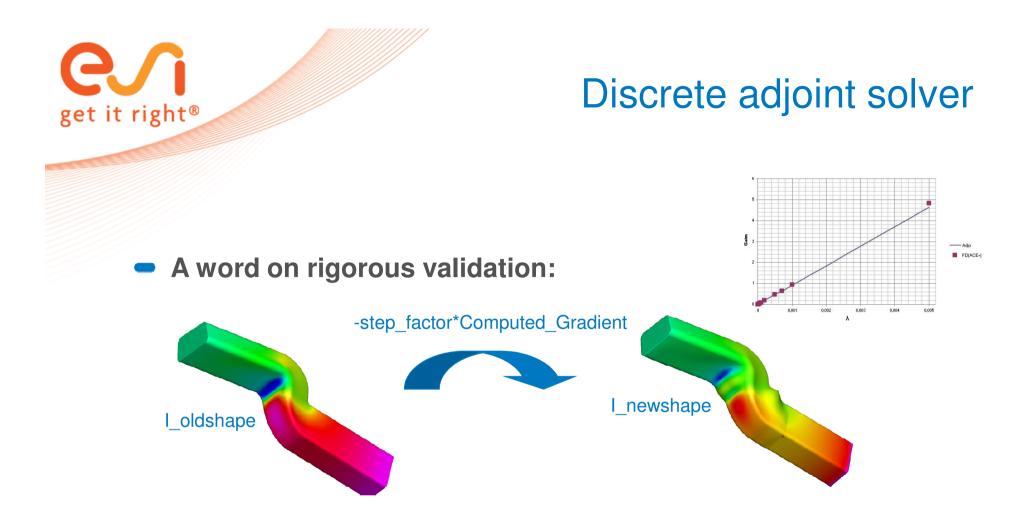
Dynamic library interfaced with CFD-ACE+

Any mesh topology

- Enriched with converters so that it can accomodate results of alternative CFD codes
 - the CFD may be run using OpenFOAM or Star-CCM+ and the adjoint using PAM-FLOW

Limitations:

Not yet parallelized



Expected Improvement := step_factor*Computed_Gradient_Norm**2

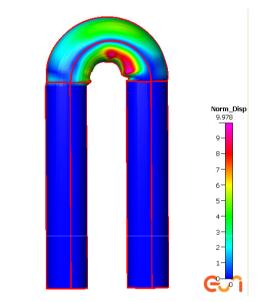
Effective Improvement := I_oldshape-I_newshape

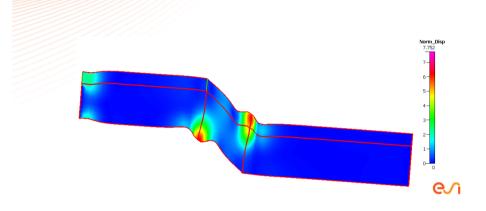
Relative error :=100* ABS(Expected_Improvement-Effective_Improvement) /Effective_Improvement

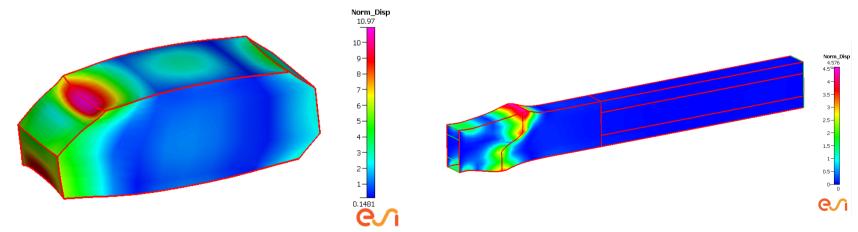


| | Airfoil Tet | Airfoil Hex | Aifoil Hex | S-Bend (Viscart) | Ahmed Body (Viscart) |
|-----------------------|----------------|----------------|---------------------|---------------------|----------------------------|
| Physics | Laminar | Laminar | Frozen Turbulent | Frozen Turbulent | Frozen Turbulent |
| Relative error (%) | 0,07 | 0,14 | 0,37 | 5,15 | 0,11 |









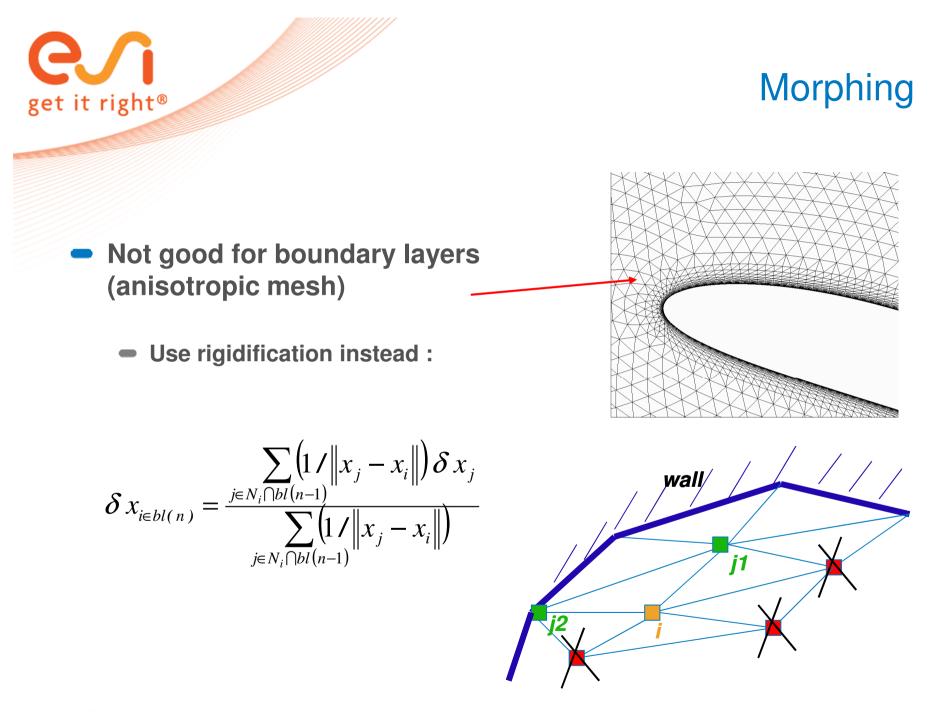




 At first bound interior node displacement to sruface nodes one thanks to harmonic mapping (ALE-like):

Minimizes mesh isotropic distortion

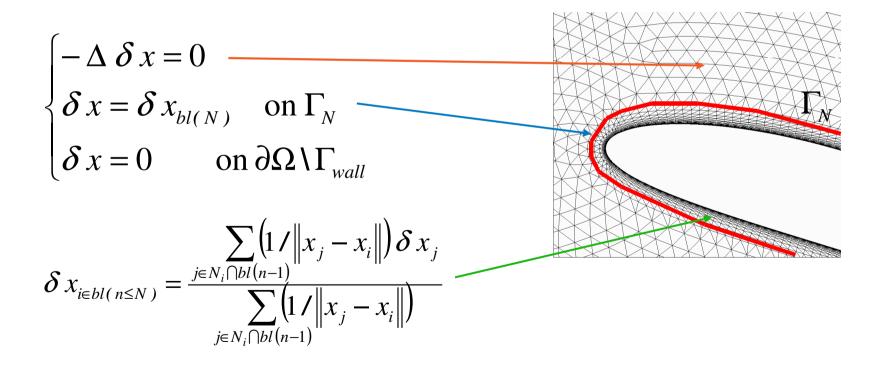
$$\begin{cases} -\Delta \,\delta \, x = 0 \\ \delta \, x = \delta \, x_{surf} & \text{on } \Gamma_{wall} \\ \delta \, x = 0 & \text{on } \partial \Omega \setminus \Gamma_{wall} \end{cases} \longrightarrow \delta \, x_{int} = K \, \delta \, x_{surf} \end{cases}$$





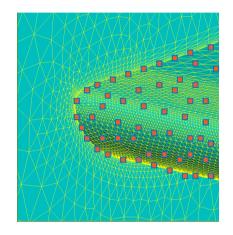
Morphing

Then combine Laplace and rigidification:



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Morphing



For boundary nodes, apply LSQ morphing:

- Sample the surface nodes $(\delta x_{surf}^i) \rightarrow (\delta y_s^j)$
- Apply LSQ operator (exact for polynomial up to desired degree)

$$\delta x_{surf}(x) = \sum_{j \in S} \delta y_{s}^{j} \Phi_{j}(x)$$

$$(\Phi_{i}(x))_{1 \leq i \leq n} = \operatorname{argmin}\left(J(\lambda) = \sum_{i \in S} W_{\varepsilon}(x - x_{i}) \lambda_{i}^{2}\right)$$
subject to
$$\sum_{i node} \lambda_{i} p_{k}(x_{i}) = p_{k}(x) \quad \forall p_{k}$$

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Finally, adapt the shape derivative accordingly via chain rule:

$$\frac{\partial \ell}{\partial y_s} = \left\{ \left(\frac{\partial \ell}{\partial x_{in}} \right) \left(\frac{\partial x_{in}}{\partial x_{surf}} \right) + \left(\frac{\partial \ell}{\partial x_{surf}} \right) \right\} \left(\frac{\partial x_{surf}}{\partial y_s} \right)$$

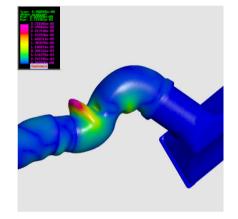
Do not forget this step, otherwise the gradient is wrong !

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Morphing

 Morphing option available both within PAM-FLOW Continuous Adjoint Solver and ACE+ Discrete one



Limited to small displacements

 Additionaly, the tool provides the value of the maximal step factor so that all volume remain positive



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